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METRIC Designation: B 221M – 0<del>0</del>2

# Standard Specification for Aluminum and Aluminum-Alloy Extruded Bars, Rods, Wire, Profiles, and Tubes [Metric]<sup>1</sup>

This standard is issued under the fixed designation B 221M; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon ( $\epsilon$ ) indicates an editorial change since the last revision or reapproval.

This standard has been approved for use by agencies of the Department of Defense.

<sup>1</sup> This specification is under the jurisdiction of ASTM Committee B07 on Light Metals and Alloys and is the direct responsibility of Subcommittee B07.03 on Aluminum Alloy Wrought Products.

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#### 1. Scope \*

1.1 This specification covers aluminum and aluminum-alloy extruded bar, rod, wire, profile, and tube in the aluminum alloys (Note 1) and tempers shown in Table 2.

Note 1-Throughout this specification the use of the term alloy in the general sense includes aluminum as well as aluminum alloy.

NOTE 2—For rolled or cold-finished bars and rods refer to Specification B 211M, for drawn tube to Specification B 210M, for structural pipe and tube to Specification B 429M, and for seamless pipe and tube to Specification B 241/B 241M.

1.2 Alloy and temper designations are in accordance with ANSI H35.1M. The equivalent Unified Numbering System alloy designations are those of Table 1 preceded by A9, for example, A91100 for Aluminum 1100 in accordance with Practice E 527.

1.3 For acceptance criteria for inclusion of new aluminum and aluminum alloys in this specification, see Annex A2.

1.4 This specification is the metric counterpart of Specification B 221.

#### 2. Referenced Documents

2.1 The following documents of the issue in effect on the date of material purchase form a part of this specification to the extent referenced herein:

2.2 ASTM Standards:

- B 557M Test Methods of Tension Testing Wrought 210M Specification for Aluminum and Cast Aluminum- and Magnesium-Alloy Products Aluminum-Alloy Drawn Seamless Tubes [Metric]<sup>2</sup>
- B-594 Practice 211M Specification for Ultrasonic Inspection of Aluminum and Aluminum-Alloy Wrought Products for Aerospace Applications Bar, Rod, and Wire [Metric]<sup>2</sup>
- B-597 Practice 241/B 241M Specification for Heat Treatment of Aluminum and Aluminum-Alloy Seamless Pipe and Seamless Extruded Tube [Metric]<sup>2</sup>
- B 429M Specification for Aluminum-Alloy Extruded Structural Pipe and Tube [Metric]<sup>2</sup>
- B 557M Test Methods of Tension Testing Wrought and Cast Aluminum- and Magnesium-Alloy Products [Metric]<sup>2</sup>
- B 594 Practice for Ultrasonic Inspection of Aluminum-Alloy Wrought Products for Aerospace Applications<sup>2</sup>
- <u>B</u>660 Practices for Packaging/Packing of Aluminum and Magnesium Products<sup>2</sup>
- B 666/B 666M Practice for Identification Marking of Aluminum and Magnesium Products<sup>2</sup>
- B 807 Practice for Extrusion Press Solution Heat Treatment of Aluminum Alloys<sup>2</sup>
- <u>B 918 Practice for Heat Treatment of Wrought Aluminum Alloys<sup>2</sup></u>
- E 29 Practice for Using Significant Digits in Test Data to Determine Conformance with Specifications<sup>3</sup>
- E 34 Test Methods for Chemical Analysis of Aluminum and Aluminum-Base Alloys<sup>4</sup>
- E 55 Practice for Sampling Wrought Nonferrous Metals and Alloys for Determination of Chemical Composition<sup>4</sup>
- E 227 Test Method for Optical Emission Spectrometric Analysis of Aluminum and Aluminum Alloys by the Point-to-Plane Technique<sup>4</sup>

<sup>&</sup>lt;sup>2</sup> Annual Book of ASTM Standards, Vol 02.02.

<sup>&</sup>lt;sup>3</sup> Annual Book of ASTM Standards, Vol 14.02.

<sup>&</sup>lt;sup>4</sup> Annual Book of ASTM Standards, Vol 03.05.

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 TABLE 1 Chemical Composition Limits<sup>A,B,C</sup>

Alloy	Silicon	Iron	Copper	Manga-	Magne-	Chro-	Zinc	Vanadium		Titanium	Ot Elem	her ents <sup>D</sup>	Alumi- num
				Hese	Sium	mum					Each	Total <sup>E</sup>	
1060	0.25	0.35	0.05	0.03	0.03		0.05	0.05		0.03	0.03		99.60 min <sup>F</sup>
1100	0.95 Si	+ Fe	0.05-0.20	0.05			0.10				0.05	0.15	99.00 min <sup>F</sup>
2014	0.50-1.2	0.7	3.9–5.0	0.40-1.2	0.20-0.8	0.10	0.25			0.15	0.05	0.15	remainder
2024	0.50	0.50	3.8-4.9	0.30-0.9	1.2–1.8	0.10	0.25			0.15	0.05	0.15	remainder
2219	0.20	0.30	5.8–6.8	0.30-0.40	0.02		0.10	0.05-0.15	0.10-0.25 Zr	0.02-0.10	0.05	0.15	remainder
3003	0.6	0.7	0.05-0.20	1.0–1.5			0.10				0.05	0.15	remainder
Alclad 3003		3003 c	lad with 7072	alloy									
3004	0.30	0.7	0.25	1.0–1.5	0.8–1.3		0.25				0.05	0.15	remainder
3102	0.40	0.7	0.10	0.05-0.40			0.30			0.10	0.05	0.15	remainder
5052	0.25	0.40	0.10	0.10	2.2–2.8	0.15–0.35	0.10				0.05	0.15	remainder
5083	0.40	0.40	0.10	0.40-1.0	4.0-4.9	0.05-0.25	0.25			0.15	0.05	0.15	remainder
5086	0.40	0.50	0.10	0.20-0.7	3.5–4.5	0.05-0.25	0.25			0.15	0.05	0.15	remainder
5154	0.25	0.40	0.10	0.10	3.1–3.9	0.15–0.35	0.20			0.20	0.05	0.15	remainder
5454	0.25	0.40	0.10	0.50–1.0	2.4–3.0	0.05-0.20	0.25			0.20	0.05	0.15	remainder
5456	0.25	0.40	0.10	0.50–1.0	4.7–5.5	0.05-0.20	0.25			0.20	0.05	0.15	remainder
6005	0.6–0.9	0.35	0.10	0.10	0.40-0.6	0.10	0.10			0.10	0.05	0.15	remainder
6005A	0.50-0.9	0.35	0.30	0.50 <sup>G</sup>	0.40-0.7	0.30 <sup>G</sup>	0.20			0.10	0.05	0.15	remainder
6060	0.30-0.6	0.10-0.30	0.10	0.10	0.35–0.6	0.05	0.15			0.10	0.05	0.15	remainder
6061	0.40-0.8	0.7	0.15–0.40	0.15	0.8–1.2	0.04–0.35	0.25			0.15	0.05	0.15	remainder
6063	0.20-0.6	0.35	0.10	0.10	0.45-0.9	0.10	0.10			0.10	0.05	0.15	remainder
6066	0.9–1.8	0.50	0.7–1.2	0.6–1.1	0.8–1.4	0.40	0.25			0.20	0.05	0.15	remainder
6070	1.0–1.7	0.50	0.15–0.40	0.40-1.0	0.50–1.2	0.10	0.25			0.15	0.05	0.15	remainder
6105	0.6–1.0	0.35	0.10	0.10	0.45–0.8	0.10	0.10			0.10	0.05	0.15	remainder
6162	0.40-0.8	0.50	0.20	0.10	0.7–1.1	0.10	0.25			0.10	0.05	0.15	remainder
6262	0.40-0.8	0.7	0.15-0.40	0.15	0.8–1.2	0.04-0.14	0.25		<sup>H</sup>	0.15	0.05	0.15	remainder
6351	0.7–1.3	0.50	0.10	0.40-0.8	0.40-0.8		0.20			0.20	0.05	0.15	remainder
6463	0.20-0.6	0.15	0.20	0.05	0.45-0.9		0.05				0.05	0.15	remainder
7005	0.35	0.40	0.10	0.20-0.7	1.0–1.8	0.06-0.20	4.0-5.0		0.08–0.20 Zr	0.01-0.06	0.05	0.15	remainder
7072′	0.7 Si +	Fe	0.10	0.10	0.10		0.8–1.3				0.05	0.15	remainder
7075	0.40	0.50	1.2-2.0	0.30	2.1–2.9	0.18-0.28	5.1–6.1			0.20	0.05	0.15	remainder
7116	0.15	0.30	0.50-1.1	0.05	0.8–1.4		4.2–5.2	0.05	0.03Ga	0.05	0.05	0.15	remainder
7129	0.15	0.30	0.50-0.9	0.10	1.3–2.0	0.10	4.2–5.2	0.05	0.03Ga	0.05	0.05	0.15	remainder
7178	0.40	0.50	1.6–2.4	0.30	2.4–3.1	0.18-0.28	6.3–7.3			0.20	0.05	0.15	remainder

<sup>A</sup> Limits are in mass percent maximum unless shown as a range, or stated otherwise.

<sup>B</sup> Analysis shall be made for the elements for which limits are shown in this table.

<sup>C</sup> For the purpose of determining conformance to these limits, an observed value or a calculated value obtained from analysis shall be rounded to the nearest unit in the last right-hand place of the figures used in expressing the specified limits, in accordance with the rounding-off method of Practice E 29.

<sup>D</sup>Others includes listed elements for which no specific limit is shown as well as unlisted metallic elements. The producer may analyze samples for trace elements not specified in the specification. However, such analysis is not required and may not cover all metallic Others elements. Should any analysis by the producer or the purchaser establish that an Others element exceeds the limit of Each or that the aggregate of several Others elements exceeds the limit of Total, the material shall be considered nonconforming.

<sup>E</sup>Other elements—Total shall be the sum of unspecified metallic elements 0.010 % or more, rounded to the second decimal before determining the sum.

<sup>F</sup> The aluminum content shall be calculated by subtracting from 100.00 % the sum of all metallic elements present in amounts of 0.010 % or more each, rounded to the second decimal before determining the sum.

<sup>G</sup> Manganese plus chromium shall total 0.12–0.50.

<sup>*H*</sup> Bismuth and lead shall be 0.40–0.7 % each.

<sup>1</sup> Composition of cladding alloy applied during the course of manufacture. Samples from finished tube shall not be required to conform to these limits.

E 527 Practice for Numbering Metals and Alloys (UNS)<sup>5</sup>

E 607 Test Method for <u>Optical Atomic</u> Emission Spectrometric Analysis of Aluminum and Aluminum Alloys by the Point-to-Plane Technique, Nitrogen Atmosphere<sup>4</sup>

E 716 Practices for Sampling Aluminum and Aluminum Alloys for Spectrochemical Analysis<sup>4</sup>

E-1004 Test Method 1004 Practice for Electromagnetic (Eddy-Current) Measurements of Determining Electrical Conductivity Using the Electromagnetic (Eddy-Current) Method<sup>6</sup>

E 1251 Test Method for Optical Emission Spectrometric Analysis of Aluminum and Aluminum Alloys by the Argon Atmosphere, Point-to-Plane, Unipolar Self-Initiating Capacitor Discharge<sup>6\_4</sup>

G 47 Test Method for Determining Susceptibility to Stress-Corrosion Cracking of High-Strength 2xxx and 7xxx Aluminum Alloy Products<sup>7</sup>

Method of Test for Exfoliation Corrosion Susceptibility in 7xxx Series Copper-Containing Aluminum Alloys (EXCO Test) (G34-72)<sup>8</sup>

2.3 ANSI Standards:

<sup>&</sup>lt;sup>5</sup> Annual Book of ASTM Standards, Vol 01.01.

<sup>&</sup>lt;sup>6</sup> Annual Book of ASTM Standards, Vol 03.063.

<sup>&</sup>lt;sup>7</sup> Annual Book of ASTM Standards, Vol 03.032.

<sup>&</sup>lt;sup>8</sup> The applicable edition in the use of this specification is G 34–72, which is available in the gray pages of the Annual Book of ASTM Standards, Vol 032.02.

H35.1(M) Alloy and Temper Designation Systems for Aluminum<sup>2</sup>

H35.2(M) Dimensional Tolerances for Aluminum Mill Products<sup>2</sup>

2.4 ISO Standards:9

ISO 209-1 Wrought Aluminum and Aluminum Alloys-Chemical Composition and Forms of Product

ISO 2107 Aluminum, Magnesium and their Alloys-Temper Designation

ISO 6362-2 Wrought Aluminum and Aluminum Alloy Extruded Rod/Bar, Tube, and Profile—Mechanical Properties

2.5 Federal Standard:

Fed. Std. No. 123 Marking for Shipment (Civil Agencies)<sup>10</sup>

2.6 *Military Standard:* 

MIL-STD-129 Marking for Shipment and Storage<sup>10</sup>

2.7 *Military*AMS Specification:

MIL-H-6088 Heat

AMS 2772 Heat Treatment of Aluminum Alloy Raw Materials<sup>11</sup>

#### 3. Terminology

3.1 *Definitions:* 

3.1.1 *extruded bar*—an extruded solid section, long in relation to its cross-sectional dimensions, having asymmetrical cross section that is square or rectangular with sharp or rounded corners or edges, or is a regular hexagon or octagon, and whose width or greatest distance between parallel faces is over 10 mm.

3.1.2 extruded rod—an extruded round section, long in relation to its diameter, whose diameter is over 10 mm.

3.1.3 *extruded profile*—a hollow or solid extruded section, long in relation to its cross-sectional dimensions, whose cross section is other than that of wire, rod, bar, or tube.

3.1.4 *extruded tube*—an extruded hollow section, long in relation to its cross-sectional dimensions, which is symmetrical and is round, square, rectangular, hexagonal, or elliptical with sharp or rounded corners, and has a uniform wall thickness except as affected by corner radii.

3.1.5 *alclad tube*—tube having on the inside surface a metallurgically bonded aluminum or aluminum-alloy coating that is anodic to the core alloy to which it is bonded, thus electrolytically protecting the core alloy against corrosion.

3.1.6 *wire*—a solid section long in relation to its cross-sectional dimensions, having a cross section that is round, hexagonal, or octagonal and whose diameter, width, or greatest distance between parallel faces is up through 10 mm, or having a symmetrical cross section that is square or rectangular (excluding flattened wire) with sharp or rounded corners or edges.

3.1.7 producer—the primary manufacturer of a material.

3.1.8 supplier-includes only the category of jobbers and distributors as distinct from producer.

3.2 Definitions of Terms Specific to This Standard:

3.2.1 *capable of*—The term *capable of* as used in this specification means that the test need not be performed by the producer of the material. However, should subsequent testing by the purchaser establish that the material does not meet these requirements, the material shall be subject to rejection.

<sup>9</sup> The applicable edition in the use of this specification is G 34–72, which is available in the gray pages of the Annual Book of ASTM

<sup>9</sup> Available from American National Standards, Vol 02.02. Institute (ANSI), 25 W. 43rd St., 4th Floor, New York, NY 10036.

<sup>10</sup> Available from American National Standards Institute, 11 W. 42nd St., 13th Floor, New York, NY 10036. <u>Standardization Documents Order Desk, Bldg. 4 Section D</u>, 700 Robbins Ave., Philadelphia, PA 19111-5094, Attn: NPODS.

<sup>&</sup>lt;sup>11</sup> Available from Standardization Documents Order Desk, Bldg. 4 Section D, 700 Robbins Ave., Philadelphia, Society of Automotive Engineers (SAE), 400 Commonwealth Dr., Warrendale, PA-19111-5094, Attn: NPODS. 15096-0001.

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TABLE 2 Tensile Property Limits<sup>A,B</sup>

	Specified Section <u>Thickness</u> ,	on or Wall mm	<u>Area,</u>	mm <sup>2</sup>	Tensile Stre	ength, MPa	Yield Stren offset)	igth (0.2 % , MPa	Elonga	ation, <sup><i>C</i></sup> %, min
Temper	over	incl	over	incl	min	max	min	max	<u>in 50 mm</u>	$\frac{\text{in 5} \times}{\text{Diameter}}$ (5.65 $\sqrt{A}$ )
					Aluminum 10	<u> 50</u>				<u> </u>
$\frac{\underline{O}}{\underline{H112}}$ $\underline{\underline{F}^{D}}$	all all all		all all all		<u>60</u> <u>60</u> 	<u>95</u> 	<u>15</u> <u>15</u> 	· · · · · · · ·	25 25 	<u>22</u> 22 
					Aluminum 11	00				
<u>O</u> <u>H112</u> <u>F<sup>D</sup></u>	all all all		all all all		<u>75</u> <u>75</u> 	<u>105</u>  	<u>20</u> <u>20</u> 	···· ····	25 25 	22 22 
					Alloy 2014					
<u>O</u> <u>T4</u> <u>T4510<sup>E</sup> T4511<sup>E</sup></u>	} all all		all all		<u></u> <u>345</u>	<u>205</u> 	<u>240</u>	<u>125</u> 	<u>12</u> 12	<u>10</u> <u>10</u>
$\frac{14311}{T42^{F}}$ $\frac{1}{T6}$ $\frac{1}{T6510^{E}}$ $\frac{1}{T6511^{E}}$ $\frac{1}{T62^{F}}$	$ \begin{array}{c} \begin{array}{c} \begin{array}{c} \begin{array}{c} \begin{array}{c} \begin{array}{c} \begin{array}{c} \begin{array}{c}$	<u>12.50</u> <u>18.00</u> <u></u> <u>18.00</u> <u></u>	all all <u>16 000</u> all <u>16 000</u> all	<u>16 000</u> <u>20 000</u> <u>16 000</u> <u>20 000</u> 	$     \begin{array}{r} 345 \\             415 \\             440 \\             470 \\             415 \\             415 \\             415 \\             415 \\             415 \\          $		$     \begin{array}{r}       200 \\       365 \\       400 \\       415 \\       400 \\       365 \\       $		12 7  7 	10 6 6 6 5 6 5 
					Alloy 2024					
$\frac{O}{T3} \\ \overline{T3510^{E}} \\ \overline{T3511^{E}} \\ \overline{T42^{F}} \\ \frac{T42^{F}}{\overline{T8510^{E}}} \\ \overline{T8510^{E}} \\ \overline{T8511^{E}} \\ \overline{F^{D}} \\ -$	$\begin{array}{c} \underbrace{all} \\ \hline \\ \hline \\ \hline \\ \hline \\ \hline \\ \\ \\ \hline \\ \hline \\ \\ \hline \\ \\ \hline \\ \hline \\ \\ \hline \hline $	6.30 18.00 35.00  18.00 35.00  6.30 35.00 	all       all       all       all       16 000       all       all       all       all       all       all       all       all       all	$     \frac{16\ 000}{20\ 000}     \dots     16\ 000}{20\ 000}     20\ 000   $	 395 415 450 485 450 395 395 395 395 395 395 395 440 455 455 455	240     	$\begin{array}{c} & \ddots \\ & 290 \\ \hline 305 \\ & 315 \\ \hline 360' \\ \hline 260 \\ \hline 400 \\ \hline 400 \\ \hline \cdots \end{array}$		$ \begin{array}{c} \underline{12}\\ \underline{12}^{G}\\ \underline{12}^{G}\\ \vdots\\ \vdots\\ \underline{12}^{G}\\ \underline{12}^{G}\\ \vdots\\ \underline{12}\\ 12$	$   \begin{array}{c}     10 \\     \overline{10^{G}} \\     9 \\     9 \\     \overline{7} \\     \overline{10} \\     9 \\     \overline{9} \\     \overline{7} \\     \overline{10} \\     9 \\     \overline{9} \\     \overline{7} \\     \overline{10} \\     \overline{9} \\     \overline{9} \\     \overline{7} \\     \overline{10} \\     \overline{9} \\     \overline{9} \\     \overline{7} \\     \overline{10} \\     \overline{9} \\     \overline{9} \\     \overline{7} \\     \overline{7} \\     \overline{10} \\     \overline{9} \\     \overline{9} \\     \overline{7} \\     \overline{7} \\     \overline{10} \\     \overline{9} \\     \overline{9} \\     \overline{7} \\     \overline{7} \\     \overline{10} \\     \overline{9} \\     \overline{9} \\     \overline{7} \\     \overline{7} \\     \overline{10} \\     \overline{9} \\     \overline{9} \\     \overline{7} \\     \overline{7} \\     \overline{10} \\     \overline{9} \\     \overline{9} \\     \overline{7} \\     \overline{7} \\     \overline{7} \\     \overline{10} \\     \overline{9} \\     \overline{9} \\     \overline{7} \\      \overline{7} \\     \overline{7} $
					Alloy 2219					
<u>O</u> <u>T31</u> <u>T3510<sup>E</sup></u> <u>T62<sup>F</sup></u>	$ \begin{array}{c} \underline{all} \\ \underline{\cdot \cdot \cdot} \\ \underline{-12.50} \\ $	12.50 80.00 25.00	all 	<u>16 000</u> <u>16 000</u> <u>16 000</u> 20 000	290 310 370 370	<u>220</u>  	180 185 250 250	<u>125</u>  	<u>12</u> <u>14</u> <u></u> <u>6</u> 	$ \begin{array}{r} 10\\ 12\\ 12\\ 5\\ 5\\ 5 \end{array} $
<u>181</u> <u>T8510<sup>E</sup></u> T8511 <sup>E</sup>	<u>}</u>	80.00	<u></u>	<u>16 000</u>	400	<u></u>	<u>290</u>	<u></u>	<u>6</u>	<u>5</u>
<u>F<sup>D</sup></u>	all		all		<u></u>	<u></u>	<u></u>	<u></u>	<u></u>	<u></u>
0 H112 F <sup>D</sup>	all all all		all all all		Alloy 3003 95 90  Alclad Alloy 30	<u>130</u>  	<u>35</u> <u>30</u> 	···· ···· ····	25 25 	<u>22</u> <u>22</u> 
<u>0</u>	all	all	all		90	<u>125</u>	<u>30</u>	<u></u>	25	<u></u>
<u>H112</u>	all				<u>90</u>	<u></u>	<u>30<sup>J</sup></u>	<u></u>	<u>25</u>	<u></u>
					Alloy 3004	200	60			
<u> </u>	all		all			<u>200</u>	<u></u>	<u></u>	<u></u>	<u></u>

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TABLE 2 Continued

	Specified Sect Thicknes	tion or Wall s, mm	Area	i, mm²	Tensile Stre	ngth, MPa	Yield Strer offset)	ngth (0.2 % ), MPa	Elonga	ation, <sup><i>C</i></sup> %, min
Temper	over	incl	over	incl	min	max	min	max	<u>in 50 mm</u>	$\frac{in 5 \times}{Diameter}$ (5.65 $\sqrt{A}$ )
					Alloy 3102					<u> </u>
H112 <sup><i>K</i></sup>	0.70	1.30	all		75	125	30	<u></u>	25	<u></u>
					Alloy 5052					
<u>0</u>	all		all		170	240	<u>70</u>	<u></u>	<u></u>	<u></u>
					Alloy 5083					
0	<u></u>	130.00 <sup>L</sup>	<u></u>	20 000	270	350	110	<u></u>	14	<u>12</u>
H111 H112	<u></u>	$\frac{130.00^{L}}{130.00^{L}}$	<u></u>	<u>20 000</u> 20 000	$\frac{275}{270}$	<u></u>	<u>165</u> 110	<u></u>	<u>12</u> 12	$\frac{10}{10}$
$\overline{F^{\mathcal{D}}}$	all		all		<u></u>	<u></u>	<u></u>	····	<u></u>	<u></u>
					Alloy 5086					
$\underline{O}$	<u></u>	$\frac{130.00^{L}}{130.00^{L}}$	<u></u>	20 000	240	<u>315</u>	<u>95</u>	<u></u>	$\frac{14}{12}$	$\frac{12}{10}$
H1112	<u></u>	$\frac{130.00^{-1}}{130.00^{-1}}$	<u></u>	20 000	250	<u></u>	95	<u></u>	12	10
F <sup>₽</sup>	all		all		<u></u>	<u></u>	<u></u>	<u></u>	<u></u>	<u></u>
					Alloy 5154					
<u>O</u> <u>H112</u>	all all		all all		205 205	<u>285</u>	75 75	<u></u>	<u></u>	···· ····
					Alloy 5454					
0	<u></u>	$\frac{130.00^{L}}{130.00^{L}}$	<u></u>	20 000	215	285	85	<u></u>	$\frac{14}{12}$	$\frac{12}{10}$
H111 H112	<u></u>	$\frac{130.00^{2}}{130.00^{2}}$	<u></u>	20 000	2 <u>30</u> 215	<u></u>	<u>130</u> 85	<u></u>	$\frac{12}{12}$	<u>10</u> 10
F <sup>D</sup>	all		all		····		<u></u>	····		
					Alloy 5456					
<u>O</u> ⊔111	<u></u>	$\frac{130.00^{L}}{130.00^{L}}$	<u></u>	20 000	<u>285</u> 200	<u>365</u>	<u>130</u> 180	<u></u>	$\frac{14}{12}$	$\frac{12}{10}$
H112	<u></u>	130.00 <sup>L</sup>	<u></u>	20 000	285	<u></u>	130	<u></u>	12	10
<u>F<sup>D</sup></u>	all		all		<u></u>	<u></u>	<u></u>	<u></u>	<u></u>	<u></u>
					Alloy 6005					
$\frac{T1}{T5}$	<u></u>	<u>12.50</u> 3.20	<u>all</u> all		$\frac{170}{260}$	<u></u>	<u>105</u> 240	<u></u>	<u>16</u> 8	<u>14</u> 
	3.20	25.00	all		260		240		_ 10	9
					Alloy 6005A					
T1		6.30	all		170		100		15	
<u>T5</u>	<u> </u>	6.30	all		260	<u></u>	215	····	$\frac{\overline{7}}{\overline{0}}$	 
	0.30	25.00			<u>200</u>	<u></u>	215	<u></u>	9	<u>o</u>
 T51		3 20	all		150		110		8	
	<u></u>				Allov 6061	<u></u>		<u></u>	<u>0</u>	<u></u>
0	all		all		<u>/ moy coor</u>	150		110	16	14
$\frac{1}{T1}$	<u></u>	16.00	all		180	<u></u>	95	<u></u>	16	14
$\left\{ \frac{T4}{T4510^{E}} \right\}$	all		all		<u>180</u>	<u></u>	<u>110</u>	<u></u>	<u>16</u>	<u>14</u>
T4511 <sup>E</sup>	<u>un</u>									
T42 <sup>-</sup> T51	all	16.00	<u>all</u> all		$\frac{180}{240}$	<u></u>	<u>85</u> 205	<u></u>	<u>16</u> 8	$\frac{14}{7}$
$\overline{\text{T6, T62}^F}$	( <u></u>	6.30	all		260	<u></u>	240	<u></u>	8	-
$\frac{16510^{2}}{16511^{E}}$	<u> </u>	<u></u>	all		260	<u></u>	240	<u>···</u>	<u>10</u>	9
F <sup>D</sup>	all		all		<u></u>	<u></u>	<u></u>	<u></u>	<u></u>	<u></u>
					Alloy 6063					
$\frac{O}{T_1}$	all	12 50	<u>all</u> all		<u></u> 115	130	<u></u> 60	<u></u>	<u>18</u> 12	<u>16</u> 10
<u> </u>	12.50	25.00	all	<u></u>	110	<u></u>	55	<u></u>	<u>'</u>	10
<u>T4,T42</u> <sup>F</sup>	<u></u> 12.50	<u>12.50</u> 25.00	<u>all</u> all		$\frac{130}{125}$	<u></u>	$\frac{70}{60}$	<u></u>	<u>14</u>	$\frac{12}{12}$
<u>T5</u>	····	12.50	all		150	<u></u>	110	· · · ·	8	7
	12.50	25.00	all		145		105			/

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TABLE 2 Continued

	Specified Sect Thickness	ion or Wall s, mm	Are	a, mm²	Tensile Stre	ngth, MPa	Yield Stren offset)	igth (0.2 % , MPa	Elong	ation, <sup>C</sup> %, min
Temper	over	incl	over	incl	min	max	min	max	<u>in 50 mm</u>	$\frac{\frac{\text{in 5} \times}{\text{Diameter}}}{(5.65 \sqrt{A})}$
<u>Т52</u> Т6, Т62 <sup>к</sup>	····  3.20	25.00 3.20 25.00	all all all		150 205 205	<u>205</u> 	<u>110</u> <u>170</u> <u>170</u>	<u>170</u> 	8 8 10	<u>7</u> <u></u> <u>9</u>
					Alloy 6066					
$\left. \begin{array}{c} O \\ \overline{T4} \\ \overline{T4510^{E}} \\ \overline{T4514^{E}} \end{array} \right\}$	all all		all all		275	<u>200</u> 	<u></u> 170	<u>125</u> 	<u>16</u> <u>14</u>	<u>14</u> <u>12</u>
$\begin{array}{c} \underline{14511^{E}} \\ \underline{\underline{145}} \\ \underline{\underline{145}} \\ \underline{\underline{16}} \\ \underline{\underline{16510}} \\ \underline{16510} $	<u>all</u> all		all all		<u>275</u> <u>345</u>	 	<u>165</u> 310	···· ····	<u>14</u> <u>8</u>	<u>12</u> <u>7</u>
$\frac{T6511^{E}}{T62^{F}} \xrightarrow{\bot}$	all		all		345		290		8	7
					Alloy 6070				-	<u> </u>
T6, T62	<u></u>	80.00	<u></u>	20 000	330	<u></u>	310	<u></u>	6	5
					Alloy 6105					
	<u></u>	12.5	all		<u>170</u>	<u></u>	105	<u></u>	<u>16</u>	<u>14</u>
15	3.20	25.00	all		<u>250</u> 250	<u></u>	240	<u></u>	<u>8</u> 10	<u></u> <u>9</u>
					Alloy 6162					
$\left. \begin{array}{c} \underline{T5}, \\ \overline{T5510^{E}} \\ \overline{T5511^{E}} \end{array} \right\}$	<u></u>	25.00	all		<u>255</u>	<u></u>	<u>235.</u>	<u></u>	<u>7</u>	<u>6</u>
$ \frac{\overline{16,}}{\overline{16510^{E}}} $	6.30	6.30 12.50	all all		260 260	···· ····	240 240	<u></u>	<u>8</u> <u>10</u>	<u>9</u>
					Alloy 6262					
$ \begin{array}{c} \underline{T6} \\ \underline{T6510}^{E} \\ \underline{T6511}^{E} \end{array} \end{array} \right\} $	all		all		<u>260</u>	<u></u>	<u>240</u>	<u></u>	<u>10</u>	<u>9</u>
					Alloy 6351					
T1 T11 T4 T5 T51 T54 T6	···· ···· 6.30 3.20 ···· 3.20	$ \begin{array}{r} 12.50\\ 19.00\\ \hline 19.00\\ \hline 6.30\\ \hline 25.00\\ \hline 25.00\\ \hline 12.50\\ \hline 3.20\\ \hline 18.00\\ \hline \end{array} $	. :   리  리  리  리  리  리  리  리	<u>13 000</u>	180 180 220 260 260 250 205 290 290		$     \begin{array}{r}       90 \\       110 \\       130 \\       240 \\       230 \\       140 \\       255 \\       255 \\     \end{array} $		15 16 16 8 10 10 10 8 10 10 8 10	13 14 14 
					Alloy 6463					
<u>T1</u> <u>T5</u> <u>T6,T62<sup>F</sup></u>	···· ···· <u>3.20</u>	12.50 12.50 3.20 12.50	···· ···· ····	$     \frac{13\ 000}{13\ 000} \\     \frac{13\ 000}{13\ 000} \\     \frac{13\ 000}{13\ 000} $	115 150 205 205	···· ···· ····	<u>60</u> <u>110</u> <u>170</u> <u>170</u>	· · · · · · · · · · ·	$\frac{12}{\underline{8}}\\ \underline{\frac{8}{\underline{8}}}\\ \underline{10}$	$\frac{10}{\underline{7}}$ $\frac{\underline{10}}{\underline{9}}$
					Alloy 7005					
<u>T53</u>	3.20	25.00	<u></u>	<u>16 000</u>	345	<u></u>	305	<u></u>	<u>10</u>	9
	2.00	40 50			Alloy 7116		000		0	7
15	3.20	12.50	all		<u>330</u>	<u></u>	290	<u></u>	<u>×</u>	<u> </u>
 T5T6		12 50	all		380		3/0		9	8
10, 10	<u></u>	12.00	<u>uii</u>		Allov 7075	<u></u>	<u>0+0</u>	<u></u>	3	<u>9</u>
<u>0</u>	all		all			275	<u></u>	<u>165</u>	<u>10</u>	<u>9</u>

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 TABLE 2
 Continued

	Specified Section	on or Wall	Area	mm <sup>2</sup>	Tensile Str	ength. MPa	Yield Stren	ath (0.2 %	Elong	ation. <sup>C</sup> %. min
	Thickness	mm				<u> </u>	offset)	MPa		, ,
	1110010000						01301)	, 1011 0		
Temper										in 5 $\times$
	over	incl	over	incl	min	may	min	may	in 50 mm	Diameter
	0101	11101	0001		<u></u>	max	<u></u>	max	11 00 11111	$\left(\frac{E}{E}\right)$
										$(3.03 \sqrt{A})$
T6 )	_	6 30	all		540		485		7	
TEOF	$\left(\begin{array}{c} \frac{1}{1} \\ \frac{1}{1} \\ \frac{1}{2} \\ \frac{1}{2}$	10.00			540	<u></u>	<del>400</del>	<u></u>	<del>1</del> 7	<u></u>
162	0.30	12.50			560	<u>···</u>	505	<u></u>	<u>/</u>	0
<u>T6510</u>	12.50	70.00	all		560	<u></u>	<u>495</u>	<u></u>	<u></u>	<u>6</u>
T6511 <sup>E</sup>	70.00	110.00	<u></u>	13 000	560	<u></u>	490			6
	70.00	110.00 <sup>L</sup>	13 000	20 000	540		485			5
-	-110.00	$130.00^{L}$		20,000	540		470			5
T73 )	1.60	6 30	<u></u>	13 000	470	<u></u>	400	<u></u>	7	5
175 TOF40F	$\left( \frac{1.00}{0.00} \right)$	0.30	<u></u>	13 000	470	<u></u>	400	<u></u>	$\frac{7}{2}$	<u></u>
13510-	6.30	35.00	<u></u>	16 000	485	<u></u>	420	<u>···</u>	8	<u>/</u>
<u>T3511</u> <sup>-</sup> –	{ <u>35.00</u>	70.00	<u></u>	<u>18 000</u>	475	<u></u>	<u>405</u>	<u></u>	<u></u>	<u>7</u>
	70.00	110.00 <sup>L</sup>		13 000	470		395			6
	70.00	110.00 <sup>L</sup>	13 000	20 000	450		380			6
		1 25	all		500		435		7	<u> </u>
	( <u>1</u> 25	2.20			510	<u></u>	100	<u></u>	- -	<u></u>
<b>TTO</b>	1.25	3.20		40.000	510	<u></u>	440	<u></u>	<u>/</u>	<u></u>
176	3.20	6.30	<u></u>	13 000	<u>510</u>	<u></u>	440	<u></u>	<u>/</u>	<u></u>
<u>T76510</u> <sup>E</sup> }	6.30	12.50	<u></u>	13 000	<u>515</u>	<u></u>	<u>450</u>	<u></u>	7	<u>6</u>
T76511 <sup>E</sup> 上	12.50	25.00		13 000	515		450			6
	25.00	50.00		13 000	515		450		7	6
	50.00	75.00	<u></u>	13 000	510	<u></u>	440	<u> </u>	7	-6
-	75.00	100.00	<u></u>	12 000	510	<u></u>	440	<u></u>	1 7	0
-0		100.00	<u></u>	13 000	510	<u></u>	435	<u></u>	<u>1</u>	<u>0</u>
<u>F</u>	all		all		<u></u>	<u></u>	<u></u>	<u></u>	<u></u>	<u></u>
					Alloy 7178					
0	all			20.000		275		165	10	0
<u>u</u>		1 60	<u></u>	12 000	ECE	215	525	105	10	<u>5</u>
		1.00	<u></u>	13 000	305	<u></u>	525	<u></u>	<u></u>	<u></u>
	1.60	6.30	<u></u>	13 000	580	<u></u>	525	<u></u>	<u>5</u>	<u></u>
T6 )	6.30	35.00	<u></u>	16 000	600	<u></u>	540	<u></u>	5	4
T6510 <sup>E</sup> }	35.00	60.00		16 000	595		530			4
T6511 <sup>E</sup>	35.00	60.00	16,000	20,000	580		515			4
	60.00	80.00		20,000	565	<u></u>	400	<u></u>	<u></u>	
-	<u> </u>	00.00	<u></u>	20 000	505	<u></u>	490	<u>···</u>	<u>· · ·</u>	<u>+</u>
	(	1.60	<u></u>	13 000	545	<u></u>	505	<u></u>	<u></u>	<u></u>
	1.60	6.30	<u></u>	13 000	565	<u></u>	510	<u></u>	5	<u></u>
	6.30	35.00		16 000	595		530		5	4
T62 <sup>F</sup>	35.00	60.00		16,000	595		530		-	4
	35.00	60.00	16,000	20,000	580	<u></u>	515	<u></u>	<u></u>	<u>.</u>
	35.00	00.00	10 000	20 000	500	<u></u>	100	<u></u>	<u></u>	<del>4</del>
-	60.00	80.00	<u></u>	20 000	565	<u></u>	490	<u></u>	<u></u>	4
<u>T76</u> )	3.20	6.30	<u></u>	13 000	525	<u></u>	455	<u></u>	<u>7</u>	<u></u>
T76510 <sup>E</sup> }	6.30	12.50		13 000	530		460		7	6
T76511 <sup>E</sup> ⊥	12.50	25.00		13 000	530	<u> </u>	460			6
F <sup>D</sup>	all		all		<u></u>	<u> </u>		<u> </u>	<u></u>	<u> </u>
<u> </u>					<u></u>	<u></u>	<u></u>	<u></u>	<u></u>	<u></u>

<sup>A</sup>The basis for establishment of tensile property limits is shown in Annex A1.

<sup>B</sup>To determine conformance to this specification, each value shall be rounded to the nearest 1 MPa for strength and the nearest 0.5 % for elongation, in accordance with the rounding-off method of Practice E 29.

<sup>C</sup>Elongations in 50 mm apply for shapes tested in full section and for sheet-type specimens machined from material up through 12.5 mm in thickness having parallel surfaces. Elongations in 5 × diameter (5.65  $\sqrt{A}$ ), where D and A are diameter and cross-sectional area of the specimen respectively, apply to round test specimens machined from thicknesses over 6.30. See 8.1.1 and 8.1.2 for conditions under which measurements are not required.

<sup>D</sup>No mechanical properties are specified or guaranteed.

<sup>E</sup>For stress-relieved tempers (T3510, T3511, T4510, T4511, T5510, T5511, T6510, T6511, T73510, T73511, T76510, T76511, T8510, T8511), characteristics and properties offer than those specified may differ somewhat from the corresponding characteristics and properties of material in the basic tempers.

<sup>F</sup>Material in the T42 and T62 tempers is not available from the material producers.

<sup>G</sup>Minimum elongation for tube, 10 % in 50 mm and 9 % in 5  $\times$  diameter.

<sup>H</sup>Minimum yield strength for tube, 330 MPa.

<sup>7</sup>Minimum yield strength for tube 315 MPa.

Yield strength is not applicable to tube.

KOnly in tube form.

<sup>2</sup>Properties not applicable to extruded tube over 70 mm wall thickness.

#### 4. Ordering Information

- 4.1 Orders for material to this specification shall include the following information:
- 4.1.1 This specification designation (which includes the number, the year, and the revision letter, if applicable),
- 4.1.2 Quantity in pieces of kilograms,
- 4.1.3 Alloy (Section 7 and Table 1),
- 4.1.4 Temper (Section 8 and Table 2),
- 4.1.5 Nominal cross-sectional dimensions as follows:
- 4.1.5.1 For rod and round wire-diameter,
- 4.1.5.2 For square-cornered bar and wire-depth and width,

4.1.5.3 For sharp-cornered hexagonal or octagonal bar and wire-distance across flats,

4.1.5.4 For round tube—outside or inside diameter and wall thickness,

4.1.5.5 For square or sharp-cornered tube other than round-distance across flats and wall thickness,

4.1.5.6 For round-cornered bar, profile, tube other than round, square, rectangular, hexagonal, or octagonal with sharp corners—drawing required, and

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4.1.6 Length.

4.2 Additionally, orders for material to this specification shall include the following information when required by the purchaser:

4.2.1 Whether heat treatment in accordance with Practice <u>B 597 B 918</u> is required (9.2),

4.2.2 Whether ultrasonic inspection is required (Section 17, Table 3),

4.2.3 Whether inspection or witness of inspection and tests by the purchaser's representative is required prior to material shipment (Section 18),

4.2.4 Whether certification is required (Section 22),

4.2.5 Whether marking for identification is required (Section 20), and whether marking of lot number for alloys 2014 and 2024 in the T3- and T4-type tempers and alloy 6061 in the T6-type tempers is required (20.2),

4.2.6 Whether Practices B 660 applies and, if so, the levels of preservation, packaging, and packing required (Section 21.3), and

4.2.7 Requirements for tensile property and dimensional tolerance for sizes not specifically covered (8.1.3 and 15.1.1).

#### 5. Materials and Manufacture

5.1 The products covered by this specification shall be produced by the hot extrusion method or by similar methods at the option of the producer, provided that the resulting products comply with the requirements in this specification.

#### 6. Quality Assurance

6.1 *Responsibility for Inspection and Tests*—Unless otherwise specified in the contract or purchase order, the producer is responsible for the performance of all inspection and test requirements specified herein. The producer may use his own or any other suitable facilities for the performance of the inspection and test requirements specified herein, unless disapproved by the purchaser in the order or at the time of contract signing. The purchaser shall have the right to perform any of the inspections and tests set forth in this specification where such inspections are deemed necessary to ensure that material conforms to prescribed requirements.

6.2 Lot Definition—An inspection lot shall be defined as follows:

6.2.1 For heat-treated tempers, an inspection lot shall consist of an identifiable quantity of material of the same mill form, alloy, temper, and nominal dimensions traceable to a heat-treat lot or lots, and subjected to inspection at one time.

6.2.2 For nonheat-treated tempers, an inspection lot shall consist of an identifiable quantity of material of the same mill form, alloy, temper, and nominal dimensions subjected to inspection at one time.

#### 7. Chemical Composition

7.1 *Limits*—The material shall conform to the chemical composition limits in Table 1. Conformance shall be determined by analyzing samples taken when the ingots are poured, or samples taken from the finished or semifinished product. If the chemical composition has been determined during the course of manufacture, analysis of the finished product shall not be required.

NOTE 3—It is standard practice in the United States aluminum industry to determine conformance to the chemical composition limits prior to further processing of ingots into wrought products. Due to the continuous nature of the process, it is not practical to keep a specific ingot analysis identified with a specific quantity of finished material.

TABLE 3 Ultrasonic Discontinuity Limits for Extruded Bar and

				Profiles <sup>A</sup>		
Allow		Thickne	ss, <sup><i>B</i></sup> mm	Mass max	Max Width,	Discontinuity
Alloy		over	incl	Piece, kg	Ratio	
2014 2024 2219	}	<u>12.50</u>	<u></u>	<u>300</u>	<u>10:1</u>	<u>B</u>
7075 7178	}	<u>12.50</u> 35.00	35.00	<u>300</u> 300	<u>10:1</u> 10:1	<u>B</u> A

<sup>A</sup>Discontinuities in excess of those listed in this table shall be allowed, subject to the approval of the procuring activity, if it is established that they will be removed by machining or that they are in noncritical areas.

<sup>C</sup>The discontinuity class limits are defined in Section 11 of Practice B 594.

<sup>&</sup>lt;sup>B</sup>The thickness of any element of a "profile" shall be deemed to be the smallest dimension of that element and the discontinuity class applicable to that particular thickness shall apply to that element of the profile.

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### 7.2 Number of Samples:

7.2.1 The number of samples taken for determination of chemical composition shall be as follows:

7.2.1.1 When samples are taken at the time the ingots are poured, at least one sample shall be taken for each group of ingots poured simultaneously from the same source of molten metal.

7.2.1.2 When samples are taken from the finished or semifinished product, a sample shall be taken to represent each 2000 kg, or fraction thereof, in the lot, except that not more than one sample shall be required per piece.

7.3 *Methods of Sampling*:

7.3.1 Samples for determination of chemical composition shall be taken in accordance with one of the following methods:

7.3.1.1 Samples for chemical analysis shall be taken from the material by drilling, sawing, milling, turning, or clipping a representative piece or pieces to obtain prepared sample of not less than 75 g. Sampling shall be in accordance with Practice E 55.

7.3.1.2 Sampling for spectrochemical analysis shall be in accordance with Practices E 716. Samples for other methods of analysis shall be suitable for the form of material being analyzed and the type of analytical method used.

NOTE 4—It is difficult to obtain a reliable analysis of each of the components of clad materials using material in its finished state. A reasonably accurate determination of the core composition can be made if the cladding is substantially removed prior to analysis. The cladding composition is more difficult to determine because of the relatively thin layer and because of diffusion of core elements to the cladding. The correctness of cladding alloy used can usually be verified by a combination of metallographic examination and spectrochemical analysis of the surface at several widely separated points.

7.4 *Methods of Analysis*—The determination of chemical composition shall be made in accordance with suitable chemical (Test Methods E 34) or spectrochemical (Test Methods E 227, E 607, and E 1251) methods. Other methods may be used only when no published ASTM test method is available. In case of dispute, the methods of analysis shall be agreed upon between the producer and purchaser.

#### 8. Tensile Properties of Material from the Producer

8.1 Limits—The material shall conform to the tensile property requirements specified in Table 2.

8.1.1 The elongation requirements shall not be applicable to the following:

8.1.1.1 Material of such dimensions that a standard test specimen cannot be taken in accordance with Test Methods B 557M, and of such profile that it cannot be satisfactorily tested in full section.

8.1.1.2 Material up through 1.60 mm in thickness.

8.1.1.3 Wire up through 3.20 mm in diameter.

8.1.2 The measurement for yield strength is not required for wire up through 3.20 mm in diameter.

8.1.3 Tensile property limits for sizes not covered in Table 2 shall be as agreed upon between the producer and purchaser and shall be so specified in the contract or purchase order.

8.2 Number of Specimens:

8.2.1 For material having a nominal mass up through 1.7 kg/linear m, one tension test specimen shall be taken for each 500 kg or fraction thereof in the lot.

8.2.2 For material having a nominal mass over 1.7 kg/linear m, one tension test specimen shall be taken for each 300 m or fraction thereof in the lot.

8.2.3 Other procedures for selecting samples may be employed if agreed upon between the producer or supplier and the purchaser.

8.3 Geometry of test specimens and the location in the product from which they are taken shall be as specified in Test Methods B 557M.

8.4 Test Methods—The tension tests shall be made in accordance with Test Methods B 557M.

8.5 *Retests*—When there is evidence that the test specimen is defective or is not representative of the lot of material, retesting may be performed in accordance with Section 8 and 9 of Test Methods B 557M.

#### 9. Heat Treatment

9.1 Producer and supplier heat treatment for the production of T1 and T5-type tempers shall be in accordance with Practice B 807 and for the production of T3, T4, T6, T7, and T8-type tempers, except as noted in 9.3 or otherwise specified in 9.2, shall be in accordance with <u>MIL-H-6088</u>. AMS 2772.

9.2 When specified, heat treatment for the production of T3, T4, T6, T7, and T8-type tempers shall be in accordance with Practice <u>B 597</u>. <u>B 918</u>.

9.3 Alloys 6061, 6063, 6162, 6463, and 6351 may be solution heat-treated and quenched at the extrusion press in accordance with Practice B 807 for the production of T4 and T6-type tempers, as applicable.

#### 10. Producer Confirmation of Heat-Treat Response

10.1 In addition to the requirements of Section 8, material in alloys 2014, 2024, and 6061 produced in the O or F temper (within the size limits specified in Table 2) shall, after proper solution heat treatment and natural aging for not less than 4 days at room temperature, conform to the properties specified in Table 2 for T42 temper material. The heat-treated samples may be tested prior to 4 days natural aging but if they fail to conform to the T42 temper properties, the test may be repeated after completion of 4 days natural aging without prejudice.

10.2 Alloys 2219, 7075, and 7178 material produced in the O or F temper (within the size limits specified in Table 2) shall, after proper solution heat treatment and precipitation heat treatment, conform to the properties specified in Table 2 for T62 temper material.

10.3 *Number of Specimens*—The number of specimens from each lot of O temper material and F temper material to be tested to verify conformance with 10.1 and 10.2 shall be as specified in 8.2.

#### 11. Heat Treatment and Reheat-Treatment Capability

11.1 As-received material in the O or F temper in alloys 2014, 2024, and 6061 (within the size limitations specified in Table 2 and without the imposition of cold work) shall be capable of conforming to the properties specified in Table 2 for T42 temper, upon being properly solution heat-treated and naturally aged for not less than 4 days at room temperature.

11.2 As-received material in the O and F tempers in alloys 2219, 7075, and 7178 (within the size limitations specified in Table 2 and without the imposition of cold work) shall be capable of conforming to the properties specified in Table 2 for the T62 temper, upon being properly solution and precipitation heat-treated.

11.3 Material in alloys and tempers 2014-T4, T4510, T4511, T6, T6510, and T6511, and 2024-T3, T3510, T3511, T81, T8510, and T8511 shall be capable of conforming to the properties specified in Table 2 for the T42 temper, upon being properly resolution heat-treated and naturally aged for not less than 4 days at room temperature.

NOTE 5-6061-T4, T6, T4510, T4511, T6510, and T6511 are deleted from 11.3 because experience has shown the reheat-treated material tends to develop large recrystallized grains and may fail to develop the tensile properties shown in Table 2.

11.4 Alloy 2219 in the T31, T3510, T3511, T81, T8510, and T8511 tempers, and alloys 7075 and 7178 in the T6, T651, T6510, and T6511 tempers shall be capable of conforming to the properties specified in Table 2 for the T62 temper, upon being properly resolution heat-treated and precipitation heat-treated.

11.5 Material in T3/T31, T3510, T3511, T4, T4510, and T4511 tempers shall be capable of conforming, upon being properly precipitation heat-treated, to the properties specified in Table 2 for the T81, T8510, T8511, T6, T6510, and T6511 tempers, respectively.

#### 12. Stress-Corrosion Resistance

Section Thickness mm

12.1 Alloy 7075 in the T73 and T76-type tempers and alloy 7178 in the T76-type tempers shall be capable of exhibiting no evidence of stress-corrosion cracking when subjected to the test specified in 12.2.

12.1.1 For lot-acceptance purposes, resistance to stress-corrosion cracking for each lot of material shall be established by testing the previously selected tension-test samples to the criteria shown in Table 4.

12.1.2 For surveillance purposes, each month the producer shall perform at least one test for stress corrosion resistance in accordance with 12.2 on each applicable alloytemper for each thickness range 20.00 and over produced that month. Each sample

Alloy and Temper	Electrical Conductivity, <sup>A</sup> % Level of Mechanical Properties IACS		Lot Acceptance Status
7075-T73	40.0 or greater	per specified requirements	acceptable
T73510 and T73511	38.0 through 39.9	per specified requirements and yield strength does not exceed minimum by more than 82 MPa	acceptable
	38.0 through 39.9	per specified requirements but yield strength exceeds minimum by more than 82 MPa	unacceptable <sup>B</sup>
	less than 38.0	any level	unacceptable <sup>B</sup>
7075-T76, T76510, and	38.0 or greater	per specified requirements	acceptable
T76511	36.0 through 37.9	per specified requirements	unacceptable <sup>B</sup>
	less than 36.0	any level	unacceptable <sup>B</sup>
7178-T76, T76510, and	38.0 or greater	per specified requirements	acceptable
T76511	35.0 through 37.9	per specified requirements	unacceptable <sup>B</sup>
	less than 35.0	any level	unacceptable <sup>B</sup>

<sup>A</sup> Sampling for electrical conductivity tests shall be the same as for tensile tests as specified in 8.2. Test specimens may be prepared by machining a flat, smooth surface of sufficient width for proper testing. For small sizes of tubes, a cut-out portion may be flattened and the conductivity determined on the surface. Chemical milling may be used on flat surface samples. The electrical conductivity shall be determined in accordance with Tes Pract-Micethod E 1004 in the following locations:

<sup>B</sup> When material is found to be unacceptable, it shall be reprocessed (additional precipitation heat treatment or re-solution heat treatment, stress relieving, straightening, and precipitation heat treatment, when applicable).

over	through	Location
	2.50	surface of tension sample
2.50	12.50	subsurface after removal of approximately 10 % of the thickness
12.50	40.00	subsurface at approximate center of section thickness, on a plane parallel to the longitudinal center line of the material.
40.00		subsurface on tension-test specimen surface that is closest to the center of the section thickness and on a plane parallel to the extrusion surface.

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shall be taken from material considered acceptable in accordance with the lot-acceptance criteria of Table 4. A minimum of three adjacent replicate specimens shall be taken from each sample and tested. The producer shall maintain records of all lots so tested and make them available for examination at the producer's facility.

12.2 The stress-corrosion cracking test shall be performed on material 20.00 mm and over in thickness as follows:

12.2.1 Specimens shall be stressed in tension in the short transverse direction with respect to grain flow and held at constant strain. The stress level shall be 75 % of the specified minimum yield strength for T73-type tempers and 170 MPa for T76-type tempers.

12.2.2 The stress-corrosion test shall be made in accordance with Test Method G 47.

12.2.3 There shall be no visual evidence of stress-corrosion cracking in any specimen, except that the retest provisions of 19.2 shall apply.

### 13. Exfoliation-Corrosion Resistance

13.1 Alloys 7075 and 7178 in the T76, T76510, and T76511 tempers shall be capable of exhibiting no evidence of exfoliation corrosion equivalent to or in excess of that illustrated by Category B in Fig. 2 of Test for Exfoliation Corrosion Susceptibility in 7xxx Series Copper Containing Aluminum Alloys (EXCO Test) (G 34-72) when tested in accordance with 13.1.1.

13.1.1 For surveillance purposes, each month at least one exfoliation-corrosion test shall be performed for each size range of extrusions produced during that month. The test shall be in accordance with Test for Exfoliation Corrosion Susceptibility in 7xxx Series Copper Containing Aluminum Alloys (EXCO Test) (G 34-72) on material considered acceptable in accordance with lot-acceptance criteria of Table 4. Specimens shall be selected at random and shall be, if possible, a minimum of 50 by 100 mm with the 100-mm dimension in a plane parallel to the direction of extrusion. The test location shall be in accordance with that specified in Table 4. The producer shall maintain records of all surveillance test results and make them available for examination at the producer's facility.

13.2 For lot-acceptance purposes, resistance to exfoliation corrosion for each lot of material in the alloys and tempers listed in 13.1 shall be established by testing the previously selected tension-test samples to the criteria shown in Table 4.

#### 14. Cladding

14.1 The aluminum-alloy cladding on clad tube shall comprise the inside surface (only) of the tube and its thickness shall be approximately 10 % of the total wall thickness.

14.2 When the cladding thickness is to be determined on finished tube, transverse cross sections of at least three tubes from the lot shall be polished for examination with a metallurgical microscope. Using a  $100 \times$  magnification, the cladding thickness at four points 90° apart in each sample shall be measured and the average of the 12 measurements shall be taken as the thickness. For a tube having a diameter larger than can be properly mounted for polishing and examination, the portions of the cross section polished for examination may consist of an arc about 12 mm in length.

#### **15. Dimensional Tolerances**

15.1 *Dimensions*—Variations from the specified dimensions for the type of material ordered shall not exceed the permissible variations prescribed in the tables of ANSI H35.2M (see Table 5).

15.1.1 Dimensional tolerances for sizes not covered in ANSI H35.2M shall be as agreed upon between the producer and purchaser and shall be so specified in the contract or purchase order.

15.2 *Sampling for Inspection*—Examination for dimensional conformance shall be made to ensure conformance to the tolerance specified.

#### 16. General Quality

16.1 Unless otherwise specified the extruded bar, rod, wire, profile, and tube shall be supplied in the mill finish and shall be uniform as defined by the requirements of this specification and shall be commercially sound. Any requirement not so covered is subject to negotiation between the producer and purchaser.

16.2 Each bar, rod, wire, profile, or tube shall be examined to determine conformance to this specification with respect to general quality and identification marking. On approval of the purchaser, however, the producer or the supplier may use a system of statistical quality control for such examination.

#### 17. Internal Quality

17.1 When specified by the purchaser at the time of placing the contract or order, each bar or profile over 12.50 mm in thickness or smallest dimension, in alloys 2014, 2024, 2219, 7075, and 7178 shall be tested ultrasonically in accordance with Practice B 594 to the discontinuity acceptance limits of Table 3.

#### **18. Source Inspection**

18.1 If the purchaser desires that his representative inspect or witness the inspection and testing of the material prior to shipment, such agreement shall be made by the purchaser and the producer or supplier as part of the purchase contract.

18.2 When such inspection or witness of inspection and testing is agreed upon, the producer or supplier shall afford the purchaser's representative all reasonable facilities to satisfy him that the material meets the requirements of this specification.



TABLE 5 Tables of ANSI H35.2M

Table No.	Title
10.1	Cross-Sectional Dimensions: Wire, Rod, Bar& Profiles
	Except for Profiles in T3510, T4510, T6510, T73510,
	T76510 and T8510 Tempers
10.2	Length: Wire, Rod, Bar and Profiles
10.3	Straightness: Rod, Bar and Profiles
10.4	Twist: Bar and Profiles
10.5	Flatness: Flat Surfaces
10.6	Flatness: Flat Surfaces, Hollow Profiles Except for O, T3510, T4510, T6510, T73510, T76510 and T8510 Tempers
10.7	Surface Roughness: Wire, Rod, Bar and Profiles
10.8	Contour (Curved Surfaces): Profiles Except for O, T3510, T4510, T6510, T73510, T76510 and T8510 Tempers
10.9	Squareness of Cut Ends: Wire, Rod, Bar and Profiles
10.10	Corner and Fillet Radii: Bar and Profiles
10.11	Angularity: Bar and Profiles Except for O, T3510, T4510, T6510, T73510, T76510, and T8510 Tempers
12.1	Diameter Round Tube Except for T3510, T4510, T6510, T73510, T76510 and T8510 Tempers
12.2	Width and Depth: Square, Rectangular, Hexagonal, Octagonal Tube Except for T3510, T4510, T6510, T73510, T76510 and T8510 Tempers
12.3	Wall Thickness: Round Tube
12.4	Wall Thickness: Other Than Round Tube
12.5	Length Extruded Tube
12.6	Twist: Other Than Round Tube
12.7	Straightness: Tube in Straight Lengths
12.8	Flatness: Flat Surfaces
12.9	Squareness of Cut Ends: Extruded Tube
12.10	Corner and Fillet Radii: Tube Other Than Round
12.11	Angularity: Tube Other Than Round
12.12	Surface Roughness: Extruded Tube
12.13	Dents: Extruded Tube

Inspection and tests shall be conducted so there is no unnecessary interference with the producer's or supplier's operations.

#### 19. Retest and Rejection

19.1 If any material fails to conform to all of the applicable requirements of this specification, it shall be cause for rejection of the inspection lot.

19.2 When there is evidence that a failed specimen was not representative of the inspection lot and when no other sampling plan is provided or approved by the purchaser through the contract or purchase order, at least two additional specimens shall be selected to replace each test specimen that failed. All specimens so selected for retest shall meet the requirements of the specification or the lot shall be subject to rejection.

19.3 Material in which defects are discovered subsequent to inspection may be rejected.

19.4 If material is rejected by the purchaser, the seller is responsible only for replacement of the material to the purchaser. As much as possible of the rejected material shall be returned to the seller by the purchaser.

### 20. Identification Marking of Product

20.1 When specified in the contract or purchase order, all material shall be marked in accordance with Practice B 666/B 666M. 20.2 In addition, alloys 2014, 2024, 2219, 7075, and 7178 in the T6-, T73-, T76-, and T8-type tempers and, when specified, alloys 2014, 2024, and 6061 in the T3- and T4-type tempers and alloy 6061 in the T6-type tempers shall also be marked with the lot number in at least one location on each piece.

20.3 The requirements specified in 20.1 and 20.2 are the minimum; marking systems that involve added information, larger characters, and greater frequencies are acceptable under this specification, and shall be agreed upon by the producer and purchaser.

### 21. Packaging and Package Marking

21.1 The material shall be packaged to provide adequate protection during normal handling and transportation and each package shall contain only one size, alloy, and temper of material unless otherwise agreed upon. The type of packing and gross mass of containers shall, unless otherwise agreed upon, be at the producer or supplier's discretion, provided they are such as to ensure acceptance by common or other carriers for safe transportation at the lowest rate to the delivery point.

21.2 Each shipping container shall be marked with the purchase order number, material size, specification number, alloy and temper, gross and net masses, and the producer's name or trademark.

21.3 When specified in the contract or purchase order, material shall be preserved, packaged, and packed in accordance with the requirements of Practices B 660. The applicable level shall be as specified in the contract or order. Marking for shipment of such material shall be in accordance with Fed. Std. No. 123 for civil agencies and MIL-STD-129 for Military agencies.

#### 22. Certification

22.1 The producer or supplier shall, on request, furnish to the purchaser a certificate stating that each lot has been sampled, tested, and inspected in accordance with this specification and has met the requirements.

#### 23. Keywords

23.1 aluminum alloy; extruded bars; extruded rods; extruded profiles; extruded tubes; extruded wire

#### ANNEXES

#### (Mandatory Information)

#### A1. BASIS FOR INCLUSION OF PROPERTY LIMITS

A1.1 Limits are established at a level at which a statistical evaluation of the data indicates that 99 % of the population obtained from all standard material meets the limit with 95 % confidence. For the products described, mechanical property limits for the respective size ranges are based on the analyses of at least 100 data from standard production material with no more than ten data from a given lot. All tests are performed in accordance with the appropriate ASTM test methods. For informational purposes, refer to "Statistical Aspects of Mechanical Property Assurance" in the Related Material section of the *Annual Book of ASTM Standards*, Vol 02.02. Mechanical property limits in this metric issue were derived from the inch-pound system limits that were developed under the above principles. As test data on metric dimensioned specimens are accumulated, some refinement of limits, particularly for elongations measured in 5D, can be anticipated.

#### A2. ACCEPTANCE CRITERIA FOR INCLUSION OF NEW ALUMINUM AND ALUMINUM ALLOYS IN THIS SPECIFICATION

A2.1 Prior to acceptance for inclusion in this specification, the composition of wrought or cast aluminum or aluminum alloy shall be registered in accordance with ANSI H35.1(M). The Aluminum Association<sup>12</sup> holds the Secretariat of ANSI H35 Committee<sup>12</sup> and administers the criteria and procedures for registration.

A2.2 If it is documented that the Aluminum Association could not or would not register a given composition, an alternative procedure and the criteria for acceptance shall be as follows:

A2.2.1 The designation submitted for inclusion does not utilize the same designation system as described in ANSI H35.1(M). A designation not in conflict with other designation systems or a trade name is acceptable.

A2.2.2 The aluminum or aluminum alloy has been offered for sale in commercial quantities within the prior twelve months to at least three identifiable users.

A2.2.3 The complete chemical composition limits are submitted.

A2.2.4 The composition is, in the judgment of the responsible subcommittee, significantly different from that of any other aluminum or aluminum alloy already in the specification.

A2.2.5 For codification purposes, an alloying element is any element intentionally added for any purpose other than grain refinement and for which minimum and maximum limits are specified. Unalloyed aluminum contains a minimum of 99.00 % aluminum.

A2.2.6 Standard limits for alloying elements and impurities are expressed to the following decimal places:

Less than 0.001 % 0.001 to but less than 0.01 %	0.000X 0.00X
0.01 to but less than 0.10 %	
Unalloyed aluminum made by a refining process	0.0XX
Alloys and unalloyed aluminum not made by a refining pro-	0.0X
Cess	
0.10 through 0.55 %	0.XX
(It is customary to express limits of 0.30 through 0.55 % as	
0.X0 or 0.X5.)	
Over 0.55 %	0.X, X.X, etc.
(except that combined Si + Fe limits for 99.00 % minimum	
aluminum must be expressed as 0.XX or 1.XX)	

<sup>&</sup>lt;sup>12</sup> This specification is under the jurisdiction of ASTM Committee B07 on Light Metals and Alloys and is the direct responsibility of Subcommittee B07.03 on Aluminum Alloy Wrought Products.

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A2.2.7 Standard limits for alloying elements and impurities are expressed in the following sequence: Silicon; Iron; Copper; Manganese; Magnesium; Chromium; Nickel; Zinc (Note A2.1); Titanium; Other Elements, Each; Other Elements, Total; Aluminum (Note A2.2).

NOTE A2.1—Additional specified elements having limits are inserted in alphabetical order of their chemical symbols between zinc and titanium, or are specified in footnotes.

Note A2.2—Aluminum is specified as minimum for unalloyed aluminum and as a remainder for aluminum alloys.

### APPENDIX

#### (Nonmandatory Information)

#### X1. ISO EQUIVALENTS OF ANSI ALLOYS AND TEMPERS

X1.1 International Organization for Standardization (ISO) equivalents of the ANSI alloys and tempers given in Table X1.1 and Table X1.2 are included in ISO 209-1: 1989, Part 1, Chemical Composition and ISO 2107-1983. Mechanical property limits shown in Part 2, Mechanical Properties, of ISO 6362-2: 1990 are similar to B221M but not necessarily identical.

TABLE X1.1	ISO	Equivalents	of Allo	ys in	В	221M
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		Alloys	
ANSI	ISO	ANSI	ISO
1060	AI 99.6	6005	Al SiMg
1100	Al 99.0 Cu	6005A	AI SiMg (A)
2014	Al Cu4SiMg	6060	AI MgSi
2024	AI Cu4Mg1	6061	Al Mg1SiCu
2219	Al Cu6Mn	6263	AI Mg0.7Si
3003	Al Mn1Cu	6262	AI Mg1SiPb
3004	Al Mn1Mg1	6351	Al Si1Mg0.5Mn
5052	AI Mg2.5	7005	AI Zn4.5Mg1.5Mn
5083	Al Mg4.5Mn0.7	7075	AI Zn5.5MgCu
5086	AI Mg4	7178	AI Zn7MgCu
5154	AI Mg3.5		
5454	Al Mg3Mn		
5456	Al Mg5Mn1		

TABLE X1.2 ISO	Equivalents of	Tempers in	B 221M
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Tempers			
ANSI	ISO		
F	F		
0	0		
H112	Μ		
T1	TA		
T3	TD		
Τ4	ТВ		
T5	TE		
T6	TF		
Τ7	ТМ		
T8	TH		

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## SUMMARY OF CHANGES

This section identifies the principal changes to this standard that have been incorporated since the last issue.

(1) Deleted references to Test Method E 101. Replaced Practice B 597 with Practice B 918 in 2.2, 4.2.1 and 9.2.
(2) Replaced MIL-H-6088 with AMS 2772 in 2.7 and 9.1.

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