



Designation: F 1472 – 99

Standard Specification for Wrought Titanium - 6Aluminum - 4Vanadium Alloy for Surgical Implant Applications (UNS R56400)¹

This standard is issued under the fixed designation F 1472; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ϵ) indicates an editorial change since the last revision or reapproval.

1. Scope

1.1 This specification covers the chemical, mechanical, and metallurgical requirements for wrought annealed titanium - 6aluminum -4vanadium alloy (UNS R56400) standard grade titanium alloy to be used in the manufacture of surgical implants.

1.2 The values stated in inch-pound units are to be regarded as the standard. The metric equivalents in parentheses are provided for information only.

2. Referenced Documents

2.1 ASTM Standards:

- F 136 Specification for Wrought Titanium 6Si-4V ELI Alloy for Surgical Implant Applications²
- B 265 Specification for Titanium and Titanium Alloy Strip, Sheet and Plate³
- B 348 Specification for Titanium and Titanium Alloy Billets³
- B 381 Specification for Titanium and Titanium Alloy Forgings³
- E 8 Test Methods of Tension Testing of Metallic Materials⁴
- E 120 Test Methods for Chemical Analysis of Titanium and Titanium Alloys⁵
- E 290 Test Method for Semi-Guided Bend Test for Ductility of Metallic Materials⁴
- E 527 Practice for Numbering Metals and Alloys (UNS)⁶
- E 1409 Test Method for Determination of Oxygen in Titanium and Titanium Alloys by the Inert Gas Fusion Technique⁵
- E 1447 Test Method for Determination of Hydrogen in Titanium Alloys by the Inert Gas Fusion Thermal Conductivity Method⁷

E 1447 Test Method for Determination of Hydrogen in Titanium Alloys by the Inert Gas Fusion Thermal Conductivity Method⁷

2.2 Other Standards:

ASQC C1 Specifications of General Requirements for a

Quality Control Program⁸

2.3 Aerospace Material Specifications:

AMS 4928L Titanium Alloy Bars, Forgings and Rings 6Al-4V⁹

AMS 2249C Chemical Check Analysis Limits, Titanium and Titanium Alloys⁹

2.4 Society of Automotive Engineers Standard:

SAE J1086 Practice for Numbering Metals and Alloys (UNS)⁹

3. Product Classification

3.1 *strip*—any product under 0.1875 in. (4.76 mm in thickness and under 24 in. (610 mm) wide.

3.2 *sheet*—any product under 0.1875 in. (4.76 mm) in thickness and 24 in. (610 mm) or more in width.

3.3 *plate*—any product 0.1875 in. (4.76 mm) thick and over and 10 in. (254 mm) wide and over, with widths greater than five times thickness. Plate up to 4.00 in. (101.60 mm), thick inclusive is covered by this specification.

3.4 *bar*—rounds from 0.1875 in. (7.9 mm) to 4.00 in. (101.60 mm) in diameter. (Other sizes by special order.)

3.5 *forging bar*—bar as described in 3.4, used for production of forgings, may be furnished in the hot rolled condition.

3.6 *wire*—rounds less than 0.1875 in. (4.76 mm) in diameter.

4. Ordering Information

4.1 Inquiries and orders for material under this specification shall include the following information:

- 4.1.1 Quantity (weight or number of pieces),
- 4.1.2 Applicable ASTM designation,
- 4.1.3 Form (sheet, strip, plate, wire, bar, or forging),
- 4.1.4 Condition (see 5.1),
- 4.1.5 Mechanical properties (if applicable, for special conditions),
- 4.1.6 Finish (see 5.2),
- 4.1.7 Applicable dimensions including size, thickness, width, or print number,
- 4.1.8 Special tests, and
- 4.1.9 Special requirements

⁸ Available from American Society for Quality Control, 161 W. Wisconsin Ave., Milwaukee, WI 53203.

⁹ Available from the Society of Automotive Engineers, 3001 W. Big Beaver, Troy, MI 48084.

¹ This specification is under the jurisdiction of ASTM Committee F-4 on Medical and Surgical Materials and Devices and is the direct responsibility of Subcommittee F04.02 on Resources.

Current edition approved Feb. 10, 1999. Published June 1999. Originally published as F1472-93. Last previous edition F1472-93.

² Annual Book of ASTM Standards, Vol 13.01.

³ Annual Book of ASTM Standards, Vol 02.04.

⁴ Annual Book of ASTM Standards, Vol 03.01.

⁵ Annual Book of ASTM Standards, Vol 03.05.

⁶ Annual Book of ASTM Standards, Vol 01.01.

⁷ Annual Book of ASTM Standards, Vol 03.06.

5. Materials and Manufacture

5.1 The various titanium mill products covered in this specification are normally formed with the conventional forging and rolling equipment found in primary ferrous and non-ferrous plants. The ingot metal for such mill operations is usually melted in non consumable, plasma, EB or arc furnaces of a type conventionally used for reactive metals.

5.2 *Finish*—Annealed material may be furnished to the implant manufacturer as descaled or pickled, sandblasted, ground, machined, vapor-blasted, or combinations of these operations.

5.3 Alloy shall be multiple melted; at least one of the melting cycles shall be under vacuum. The first melt shall be made by consumable electrode, nonconsumable electrode, electron beam, or plasma arc melting practice. The subsequent melt or melts shall be made using consumable electrode practice with no alloy additions permitted in the last consumable electrode melt.

5.3.1 The atmosphere for nonconsumable electrode melting shall be inert gas at a pressure not higher than 250 mm of mercury.

5.3.2 The electrode tip for nonconsumable electrode melting shall be water-cooled copper.

5.3.3 The product, as received by the purchaser, shall be uniform in quality and condition, sound, and free from foreign materials and from internal and external imperfections detrimental to usage of the product.

6. Chemical Requirements

6.1 The heat analysis shall conform to the chemical composition of Table 1. Ingot analysis may be used for reporting all chemical requirements, except hydrogen. Samples for hydrogen shall be taken from the finished mill product.

6.2 The product analysis tolerances shall conform to the check tolerances of Table 2.

6.3 For referee purposes, Test Methods E 120, E 1409, and E 1447 shall be used.

6.4 Samples for chemical analysis shall be representative of the material being tested. The utmost care must be used in sampling titanium for chemical analysis because of its affinity for elements such as oxygen, nitrogen, and hydrogen. Therefore, in cutting samples for analysis, the operation should be carried out insofar as possible in a dust-free atmosphere. Chips should be clean and sharp. Samples for analysis should be stored in suitable containers. Cutting tools should be clean and sharp. Chips should be clean.

TABLE 1 Chemical Requirements^A

Element	Composition, %
Nitrogen, max	0.05
Carbon, max	0.08
Hydrogen, max ^B	0.015
Iron, max	0.30
Oxygen, max	0.20
Aluminum	5.5–6.75
Vanadium	3.5–4.5
Yttrium, max	0.005
Titanium ^C	Balance

^ARefer to AMS 4928 .

^BBillets shall have a maximum of 0.01 % hydrogen content.

^CThe percentage of titanium is determined by difference and need not be determined or certified.

TABLE 2 Permissible Variation in Product Analysis Tolerance^A

Element	Variation Under Min or Over Max
Nitrogen	0.02
Carbon	0.02
Hydrogen	0.002
Iron	0.10
Oxygen	0.02
Aluminum	0.04
Vanadium	0.015
Yttrium	0.0006

^ARefer to AMS 2249.

7. Mechanical Requirements

7.1 Material supplied under this specification shall conform to the mechanical property requirements given in Table 3.

7.2 Specimens for tension tests shall be machined and tested in accordance with Test Method E 8. Tensile properties shall be determined using a strain rate of 0.003 to 0.007 in./in. min (metric equivalent mm/mm min) through the specified yield strength, and then the crosshead speed shall be increased so as to produce fracture in approximately one additional minute.

7.3 For sheet and strip, the bend test specimen shall withstand being bent cold through an angle of 105° without fracture in the outside surface of the bent portion. The bend shall be made on a diameter equal to that shown in Table 3. Test condition shall conform to Test Method E 290.

8. Special Requirements

8.1 The microstructure shall be a fine dispersion of the alpha and beta phases resulting from processing in the alpha plus beta field. There shall be no continuous alpha network at prior beta grain boundaries. There shall be no coarse, elongated alpha platelets.

8.2 The beta transus temperature shall be measured by a suitable method and reported on the materials certification.

8.3 Products supplied with a machined or ground surface finish shall have no alpha case. For other products, there shall be no continuous layer of alpha case when examined at 100x.

9. Quality Program Requirements

9.1 The producer shall maintain a quality program that is defined in Specification ANSI/ASQC C1.

9.2 The manufacturer of surgical implants or medical appliances shall be assured of and may audit the producer's quality program for conformance to the intent of Specification ANSI/ASQC C1-1985, or other recognized program.

10. Marking, Packing, Certification, and Rejection

10.1 Marking, packing, certification, and rejection shall be as specified in Specifications B 265, B 348, and B 381.

11. Keywords

11.1 metals (for surgical implants); titanium alloys; orthopaedic medical devices; titanium/titanium alloys; titanium/titanium alloys (for surgical implants)

TABLE 3 Annealed Mechanical Properties^A

Nominal Diameter or Distance Between Parallel Sides, Inches (mm)	Tensile Strength psi, min (MPa)	Yield Strength at 0.2 % Offset psi, min (MPa)	Elongation ^B in 2 in. or 4D, %			Reduction of Area %, min		
			Long.	min L.T. ^C	S.T. ^C	Long.	L.T. ^C	S.T. ^C
<i>Bars and Forgings</i>								
Up to 2.0, incl. (50)	135 000 (930)	125 000 (860)	10			25	20	
Over 2.0 to 4.0, incl. (50 to 100)	130 000 (895)	120 000 (825)	10			25	20	15
Over 4.0 to 6.0, incl. (100 to 150)	130 000 (895)	120 000 (825)	10		8	20	15	
<i>Sheet Strip and Plate</i>								
Up to 0.008, excl. (0.2)	134 000 (924)	126 000 (869)						
0.008 to 0.025, excl. (0.2 to 0.6)	134 000 (924)	126 000 (869)	6					
0.025 to 0.063, excl. (0.6 to 1.6)	134 000 (924)	126 000 (869)	8					
0.063 to 0.1875, excl. (1.6 to 4.8)	134 000 (924)	126 000 (869)	10					
0.1875 to 4.00, incl. (4.8 to 101.6)	130 000 (895)	120 000 (825)	10					
Up to 0.07, incl. (1.78)								Bend Factor = 9T
Over 0.07 to 0.1875, excl. (4.8)								Bend Factor = 10T

^AMechanical properties for conditions other than those listed in this table may be established by agreement between the supplier and the implant manufacturer.

^BElongation of material 0.062 in (1.575 mm) or greater in diameter or thickness shall be measured using a gage length of 2 in. or 4D or 4W. Elongation of material under 0.062 in. (1.575 mm) in diameter or thickness may be obtained by negotiation. L = longitudinal; LT = long traverse; ST = short transverse.

^CApplies to bar, plate, and forgings only. L = longitudinal; LT = long transverse; ST = short transverse.

APPENDIXES

(Nonmandatory Information)

X1. RATIONALE

X1.1 The purpose of this specification is to characterize the chemical, physical, mechanical, and metallurgical properties of wrought annealed Ti-6Al-4V alloy to be used in the manufacture of surgical implants.

X1.2 The alloy composition covered by this standard has been employed successfully in human implants, exhibiting a well-characterized level of local biological response since 1983.(1, 2).¹⁰

X1.3 The microstructural requirements contained in this

standard represent the current general consensus of opinion with respect to optimization of mechanical properties for implant applications.

X1.4 This alloy exhibits similar mechanical properties to Specification F 136 required for the application of load-bearing orthopedic implants (2-18).

X1.5 Melting requirements specified in AMS 4928L have been included to ensure that melting practice conforms to industry guidelines for standard grade titanium - 6aluminum - 4vanadium.

¹⁰ The boldface numbers refer to references listed at the end of this standard.

X2. BIOCOMPATIBILITY

X2.1 The material composition covered by this standard has been employed successfully in contact with soft tissue and bone since 1983 (12)

material is used in appropriate applications.

X2.2 No known surgical implant has ever shown to be completely free of adverse reactions in the human body. However, long term clinical experience has shown an acceptable level of biological response can be expected, if the

X2.3 The material in this specification has been subjected to animal studies and has been shown to produce a well characterized level of biological response that is equal to or less than that produced by the reference material titanium. This material has been used clinically since 1983 (1,2,19).

REFERENCES

- (1) Dobbs, H. S., and Scales, J. T., "Behavior of Commercially Pure Titanium and Ti-318 (Ti-6Al-4V) in Orthopedic Implants," Titanium Alloys in Surgical Implants. *ASTM STP 796*, H. A. Luckey and Fred Kubli, Jr. eds., ASTM, 1983, pp. 173–186.
- (2) Dobbs, H. S., and Robertson, J. L. M., "Tensile Strength, Fatigue Life and Corrosion Behavior of Ti-318 and Ti-550," Titanium Alloys in Surgical Implants. *ASTM STP 796*, H. A. Luckey and Fred Kubli, Jr. eds., ASTM, 1983 pp. 227–237.
- (3) Semlitsch, M. F., Panic, B., Weber, H., and Schoen, R., "Comparison of the Fatigue Strength of Femoral Prosthesis Stems Made of Forged Ti-Al-V and Cobalt Base Alloys," Titanium Alloys in Surgical Implants and Materials, 1983, pp. 120–135.
- (4) Cook, S. D., Georgette, F. S., Skinner, H. B., and Haddad, R. J., Jr., "Fatigue Properties of Carbon- and Porous-Coated Ti-6Al-4V Alloys," *JBMR Vol 18*, 1984, pp. 497–512.
- (5) Yue, S., Pilliar, R. M., and Weatherly, G. C., "The Fatigue Strength of Porous-Coated Ti-6Al-4V Implant Alloy," *JBMR Vol 18*, 1984, pp. 1043–1058.
- (6) *Basic Design Facts About Titanium*, RMI Co., Niles, Ohio.
- (7) *How To Use Titanium—Properties and Fabrication of Titanium Mill Products*, Timet, Pittsburgh, Pennsylvania.
- (8) Harrigan, M. J., Haplan, M. P., and Sommer, A. W., "Effect of Chemistry and Heat Treatment on the Fracture Properties of Ti-6Al-4V Alloy," *Titanium and Titanium Alloys Source Book*, 1982, pp. 50–79.
- (9) Lewis, R. E., Bjelstich, J. G., Morton, T. M., and Crossley, F. A., "Effect of Cooling Rate on Fracture Behavior of Mill-Annealed Ti-6Al-4V," Cracks and Fracture, *ASTM STP 601*, ASTM, pp. 1976, pp. 371–390.
- (10) Chesnutt, J. C., Rhodes, C. G., and Williams, J. C., "Relationship Between Mechanical Properties, Microstructure and Fracture Topography in Alpha & Beta Titanium Alloys," Fractography Microscopic Cracking Process, *ASTM STP 600*, ASTM, pp. 1976, pp. 99–138.
- (11) Hieronymous, W. S., *Aviation Week and Space Technology*, July, 1971, p. 42.
- (12) Stubbington, C. A., and Bowen, A. W., "Improvements in the Fatigue Strength of Ti-6Al-4V through Microstructure Control," *J of Mat Sci* 9, 1974, pp. 941–947.
- (13) Yoder, G. R., Cooley, L. A., and Crooker, T. W., "A Comparison of Microstructural Effects on Fatigue—Crack Initiation and Propagation in Ti-6Al-4V," *NRL Memorandum Report*, 4758, 1982.
- (14) Bowen, A. W., and Stubbington, G. A., "The Effect of Heat Treatment on the Fatigue Strength of Ti-6Al-4V," *Titanium Science and Technology*, ed. by R. I. Joffee and H. M. Burte, Plenum Press, NY, 1973.
- (15) Bartlo, L. J., "Effect of Microstructure on the Fatigue Properties of Ti-6Al-4V Bar," Fatigue at High Temperature, *ASTM STP 459*, ASTM, 1969, pp. 144–154.
- (16) Seagle, S. R., Seeley, R. R., and Hall, G. S., "The Influence of Composition and Heat Treatment on the Aqueous-Stress Corrosion of Titanium," Applications Related Phenomena in Titanium Alloys, *ASTM STP 432*, ASTM, 1968, pp. 170–188.
- (17) Stubbington, C. A., "Metallurgical Aspects of Fatigue and Fracture in Titanium Alloys," *AGARD Proceedings*, No. 185, 1976.
- (18) *Titanium Alloys Handbook*, MCIC-HB-02 Battelle Columbus Laboratories and Wright Patterson Air Force Base, 1972, pp. 104:72–15.
- (19) Semlitsch, M., "Titanium Alloys for Hip Joint Replacements," *Clinical Materials*, 2, 1987, pp. 1–13.

The American Society for Testing and Materials takes no position respecting the validity of any patent rights asserted in connection with any item mentioned in this standard. Users of this standard are expressly advised that determination of the validity of any such patent rights, and the risk of infringement of such rights, are entirely their own responsibility.

This standard is subject to revision at any time by the responsible technical committee and must be reviewed every five years and if not revised, either reapproved or withdrawn. Your comments are invited either for revision of this standard or for additional standards and should be addressed to ASTM Headquarters. Your comments will receive careful consideration at a meeting of the responsible technical committee, which you may attend. If you feel that your comments have not received a fair hearing you should make your views known to the ASTM Committee on Standards, 100 Barr Harbor Drive, West Conshohocken, PA 19428.

This standard is copyrighted by ASTM, 100 Barr Harbor Drive, West Conshohocken, PA 19428-2959, United States. Individual reprints (single or multiple copies) of this standard may be obtained by contacting ASTM at the above address or at 610-832-9585 (phone), 610-832-9555 (fax), or service@astm.org (e-mail); or through the ASTM website (<http://www.astm.org>).